

THE ANALYSIS AND DESIGN OF BRIDGE  
MEMBERS WITH CONSTANT DEFORMATION  
FOR VARIABLE LOADING

BY

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To my wife Helen. Without her love, confidence, and patient encouragement, my graduate studies would never have been completed.

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- F.  $\text{effective\_mass} = \text{mass}/m_0$
- G.  $\text{padding\_mass} = 0.001$
- I.  $\text{cluster\_size} = \text{size}$
- L.  $\text{input\_velocity} = 0.1/\text{year}$
- M.  $\text{weight\_of\_empty\_minibin} = 0.001$
- X.  $\text{cluster\_length} = 0.1$

DISSERTATION SUBMITTED TO THE GRADUATE COUNCIL  
OF THE UNIVERSITY OF FLORIDA IN PARTIAL FULFILLMENT OF THE  
REQUIREMENTS FOR THE DEGREE OF MASTER OF PHILOSOPHY

THE STABILITY AND DESIGN OF SHOCK  
ABSORBERS WITH CONSTANT DECELERATION  
FOR VARIABLE LOADS\*

by

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December, 1958

CHAIRMAN: DR. ROSE T. WILLIAMS

Major Department: Engineering Sciences

For a given impact velocity of a vehicle striking a fixed surface a design shaped shock absorber is developed to provide the same maximum deceleration of the vehicle irrespective of the loading at impact. The spring damping and external forces are generalized. Illustrative examples are the initial part of the analysis. For a shock absorber with constant efficiency, the maximum vehicle deceleration is shown to be a function of only the impact velocity and the contact length. The same is true for spring damping and external forces. It is found to be a constant value only a function of the vehicle weight.

To illustrate the method of design synthesis, a shock absorber



using methods for finding the better parallel spring as well as  
for a number of design case examples of horizontal and vertical vehicle  
motion at impact. Stiffness area equations are developed, and dimension-  
less plots versus cushion length and closure time are presented. An  
automatic stiffness area regulator is then designed to provide optimum  
shock absorber response for all landing conditions of a freight car at  
impact. The time response of the regulated stiffness area is a function  
for the car with maximum load. At impact it is shown to be approxi-  
mately 81 of the shock absorber closure time.

## INTRODUCTION

In most industrial applications, a shock absorber is used to dissipate and/or absorb the energy of a moving mass or vehicle with a variable load [1, 16]. In practice, most shock absorbers are designed to provide a constant cushion force over the full cushion length of the shock absorber for only the maximum weight and impact velocity of the vehicle. For a partially loaded vehicle, the cushion force will have the same maximum value, but the force is no longer a constant. When the entire cushion length of the shock absorber is not utilized, the cushion deceleration of the vehicle increases, and this increase is usually undesirable.

If the full cushion length could be utilized for all vehicle loading conditions, a more efficient shock absorber could be designed to provide the optimum response of the same constant value of cushion deceleration for all velocities of the vehicle load [15, 14].

The purpose of the present dissertation is to show how such a self-regulating shock absorber can be designed to provide this conditionally better response.

## EXAMPLE

### Performance Specifications

Most products are designed to satisfy a number of requirements, or desired user, performance, reliability, maintainability goals. These requirements or specifications are usually defined and ranked in terms of any various design effort.

The performance specifications for the proposed mechanical design of a vehicle, shock absorber unit, will be as follows. The shock absorber would have a constant deceleration for all loads encountered of the vehicle at the specified maximum shock velocity. The maximum cushion length of the shock absorber is also specified.

The above specifications seem quite simple, however, in any way is possible to satisfy all of them. The analysis will determine if the simultaneous satisfaction of the specifications is possible.

## Design for Analysis

The analysis is started by creating a **Continuous** beam problem:

It will provide the desired results (e.g. the maximum stress, etc.)

1) **Length of rope,  $l$  is:**

2) **Mass per weight  $\rho = 1$  kg/m**

3) **Young's modulus  $E = 10^6$  (N/m<sup>2</sup>)**

4) **Displacement  $u = 1$  mm**

5) **Time  $t = 1$  sec**

Under **Element type**, select **111** (3D Elasticity, 8 nodes, 20 DOF).

Under **Element shape**, select **1** (Hexahedron, 20 nodes).  
It is applied to the volume:



Figure 1: New: Mesh of rope and cable system

6) **SUPPORT** under **Fixed** select **111** (3D Elasticity, 8 nodes, 20 DOF).

7) **ANALYSIS** under **Static** select **1** (Static).  
8) **POST** under **Results** select **1** (Post-Processing).  
9) **POST** under **Results** select **1** (Post-Processing).  
10) **POST** under **Results** select **1** (Post-Processing).  
11) **POST** under **Results** select **1** (Post-Processing).  
12) **POST** under **Results** select **1** (Post-Processing).

external forcing function. The classical  $\ddot{m}(x) + F(x, \dot{x})$  acting on the mass  $m$  of this generalized system will be

$$F(x, \dot{x}) = F(x, \dot{x}) - F(x, \dot{x}) - F(x, \dot{x}) \quad (1)$$

Let

$$+G(x, \dot{x}) = \text{acceleration of } m \text{ at the displacement } x$$

The reaction force will then be

$$F(x, \dot{x}) = mG(x, \dot{x}) \quad (2)$$

The kinetic energy at input plus the work done by the  $\ddot{m}(x)$  reaction function must be equal to the work absorbed or dissipated over the reaction length  $X$  of the shock absorber. Therefore

$$\frac{1}{2}mv^2 + \int_0^X F(x, \dot{x})dx = \int_0^X R(x, \dot{x})dx + \int_0^X R(x, \dot{x})dx \quad (3)$$

or

$$\frac{1}{2}mv^2 + \int_0^X F(x, \dot{x})dx = 0 \quad (4)$$

It is desired to calculate the reaction function force for  $mG(x, \dot{x})$ .

The mass  $m$  at input,  $\ddot{m}(x)$  as in a case load over the reaction length

The reaction force is considered to be optimized when it becomes zero

and the (vector) functions  $\mathbf{f}_i$  components of  $\mathbf{f}$  at node  $\mathbf{x}_i$  ( $i = 1, \dots, n$ ) are in a linear relation with all the components of  $\mathbf{u}$ :

$$\mathbf{f}_i = \mathbf{A}_i \mathbf{u} \quad (14)$$

where  $\mathbf{A}_i$  is the matrix  $\mathbf{A}$  evaluated at  $\mathbf{x}_i$ . In the  $n$  equations (14) the components of  $\mathbf{u}$  are the primary unknowns. If the matrix  $\mathbf{A}$  is constant in the domain, then the relation is constant for all nodes and it is enough to solve the system (14) for the optimal value  $\mathbf{u}^*$ .

$$\mathbf{u}^* = -\mathbf{A}^{-1} \mathbf{f} \quad (15)$$

Equation (15) may be interpreted as the equation for  $\mathbf{u}$  as a consequence of a constant length independent of  $\mathbf{u}$  (independent of the value of  $\mathbf{u}$ ). The optimal displacement is

$$\mathbf{u}^* = -\mathbf{A}^{-1} \mathbf{f} \quad (16)$$

Consequently, for all the nodes which meet  $\mathbf{u}$ , the  $\mathbf{u}$  and values of  $\mathbf{f}$  (evaluated at  $\mathbf{x}_i$ ) are:

$$\mathbf{u} = -\mathbf{A}^{-1} \mathbf{f} \quad (17)$$

$$\mathbf{f}_i = \mathbf{A}_i \mathbf{u} = -\mathbf{A}_i \mathbf{A}^{-1} \mathbf{f} \quad (18)$$

$$\ddot{y}(t) = \ddot{y}_L + \ddot{y}_R(t) \quad (10)$$

$$\ddot{y}(t) = \ddot{y}_L + \alpha \dot{y}_R(t) \quad (11)$$

The optimum position of a piston in the cylinder can therefore be defined by any one of equations (4) through (11).

To obtain this output, the cushion force must vary with the mass as shown by equation (8). Therefore, the type of damping and parallel spring must be selected with the restriction that the sum of their forces can be varied in such a manner as to optimize the cushion force for any and all vehicle landing conditions at impact. In addition, a shock absorber control will be required to adjust the cushion force, either manually or by means of a well-tuned regulator.

The need for open loop control [15, 16] requires adjustment prior to impact, for each vehicle landing condition. All open loop controls also have the disadvantage that the accuracy with which the input is adjusted determines the output. Shock absorbers with this type control are now available [17, 18], but because of the above disadvantages they have very limited application and will not be discussed further.

For the regulator or closed loop type of feedback control, the output of the shock absorber can be used to compare and actuate the cushion force for all vehicle landing conditions at impact. The block diagram of such a control is shown in Figure 3.



Figure 1: Feedback control system

When the reference signal  $r$  is a step function, the response of the system is called the step response. The step response is used to determine the system's stability and its ability to track a constant reference signal.



Figure 2: Feedback control system with a disturbance

When the reference signal  $r$  is a step function and the disturbance  $d$  is a constant, the system's response is called the step response with a disturbance. The step response with a disturbance is used to determine the system's ability to track a constant reference signal in the presence of a disturbance.





Let

then

$$\begin{aligned} \text{and } \frac{\partial}{\partial \alpha} \left( \frac{\partial \log L(\alpha)}{\partial \alpha} \right) &= \frac{\partial}{\partial \alpha} \left( \frac{\partial}{\partial \alpha} \log L(\alpha) \right) \\ &= \frac{\partial}{\partial \alpha} \left( \frac{\partial}{\partial \alpha} \log L(\alpha) \right) \\ &= \frac{\partial}{\partial \alpha} \left( \frac{\partial}{\partial \alpha} \log L(\alpha) \right) \end{aligned}$$

Let

then the derivative of  $\frac{\partial \log L(\alpha)}{\partial \alpha}$  with respect to  $\alpha$  is given by

$$\frac{\partial}{\partial \alpha} \left( \frac{\partial \log L(\alpha)}{\partial \alpha} \right) = \frac{\partial}{\partial \alpha} \left( \frac{\partial \log L(\alpha)}{\partial \alpha} \right) \quad (1)$$

then the derivative of  $\frac{\partial \log L(\alpha)}{\partial \alpha}$  with respect to  $\alpha$  is given by

$$\frac{\partial}{\partial \alpha} \left( \frac{\partial \log L(\alpha)}{\partial \alpha} \right) = \frac{\partial}{\partial \alpha} \left( \frac{\partial \log L(\alpha)}{\partial \alpha} \right) \quad (2)$$

then the derivative of  $\frac{\partial \log L(\alpha)}{\partial \alpha}$  with respect to  $\alpha$  is given by

$$\frac{\partial}{\partial \alpha} \left( \frac{\partial \log L(\alpha)}{\partial \alpha} \right) = \frac{\partial}{\partial \alpha} \left( \frac{\partial \log L(\alpha)}{\partial \alpha} \right) \quad (3)$$

then the derivative of  $\frac{\partial \log L(\alpha)}{\partial \alpha}$  with respect to  $\alpha$  is given by

the derivative of  $\frac{\partial \log L(\alpha)}{\partial \alpha}$  with respect to  $\alpha$  is given by the following expression. The expression for the derivative of  $\frac{\partial \log L(\alpha)}{\partial \alpha}$  with respect to  $\alpha$  can be obtained for any value of  $\alpha$  by substituting  $\alpha$  in the following expression

$$\frac{\partial}{\partial \alpha} \left( \frac{\partial \log L(\alpha)}{\partial \alpha} \right) = \frac{\partial}{\partial \alpha} \left( \frac{\partial \log L(\alpha)}{\partial \alpha} \right) \quad (4)$$



and the maximum rms shock absorber controls the cushion force.

The performance criteria included for any value of  $\alpha$  when either condition (1) or condition (2) is satisfied. From these equations it is possible to determine the peak is a variable which is a regulated quantity from the design process or the after design. The shock absorber force  $F_{shock}$  used as the regulated variable of the design process is shown in Figures 2 and 3.

A fixed value for  $F_{shock}$  is used as a variable is not selected for the design process. Values 150 and 180 can be used to determine the peak and cushioning process of the regulated variable that will be required. The design process of the shock absorber cushion force. Since the required response will vary with each design, the requirements for a number of specific cases which design interests will now be presented.

# auxiliary design function

where  $\beta_0$  is the

the  $\beta_0$  is the initial force is acting in the direction of the  $\beta_0$  vector, and  $\beta_0$  is the initial force is acting in the direction of the  $\beta_0$  vector, and  $\beta_0$  is the initial force is acting in the direction of the  $\beta_0$  vector.

$$\beta_0 = \beta_0 \cos \alpha_0$$

$$\beta_0 = \beta_0 \sin \alpha_0$$

where  $\beta_0$  is the

$$\beta_0 = \beta_0 \cos \alpha_0 = \frac{1}{2} \left( \frac{m_0^2 \beta_0^2}{m^2} \right)^{1/2} (1 - \alpha_0^2)^{1/2} \quad (111)$$

where  $\beta_0$  is the

$$\beta_0 = \beta_0 \sin \alpha_0 = \frac{1}{2} \left( \frac{m_0^2 \beta_0^2}{m^2} \right)^{1/2} (1 - \alpha_0^2)^{1/2} \quad (112)$$

where  $\beta_0$  is the

$$\beta_0 = \beta_0 \cos \alpha_0 = \frac{1}{2} \left( \frac{m_0^2 \beta_0^2}{m^2} \right)^{1/2} (1 - \alpha_0^2)^{1/2} \quad (113)$$

where  $\beta_0$  is the

$$\beta_0 = \beta_0 \sin \alpha_0 = \frac{1}{2} \left( \frac{m_0^2 \beta_0^2}{m^2} \right)^{1/2} (1 - \alpha_0^2)^{1/2} \quad (114)$$

When equation (111) is divided by equation (110), the equation for the displacement vector of the ellipse area is obtained

$$\frac{\beta_0 \cos \alpha_0}{\beta_0 \sin \alpha_0} = (1 - \alpha_0^2)^{1/2} \quad (115)$$

When equation (112) is divided by equation (121), the equation for the force vector in the ellipse area is obtained.

$$\frac{\beta_0 \sin \alpha_0}{\beta_0 \cos \alpha_0} = (1 - \alpha_0^2)^{1/2} \quad (116)$$

where  $\beta_0$  is the initial force is acting in the direction of the  $\beta_0$  vector. They are plotted in Figure 11.

where  $\beta_0$  is the

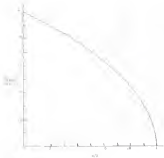


Figure 1. Relationship between normalized displacement  $u/a$  and normalized load  $P/P_0$  for a fixed area versus displacement for a fixed area.

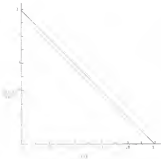


Figure 5. Relationship  $v/v_0$  vs.  $A/A_0$  for design example. Case 1.

where  $\mathbf{g}$  is the constant force is acting in the direction of motion. The potential spring is linear and has a spring constant  $k$  with no zero force

$$U(x=0) = 0$$

$$U(x) = \frac{1}{2} kx^2 \quad \text{where } k = 80 \text{ N/m}$$

constant force  $F = 10 \text{ N}$

$$U(x) = \frac{1}{2} kx^2 + \int_0^x F dx = \left( \frac{80x^2}{2} \right) + \left( \frac{1 \times x^2}{2 + 0.1x} \right) \quad (11.3)$$

$$U(x) = \left( \frac{80x^2}{2} \right) + \left[ \frac{x^2}{2 + 0.1x^2(0.2 + 0.01x)} \right] \quad (11.4)$$

$$U(x) = \frac{1}{2} kx^2 + \left( \frac{80x^2}{2} \right) \quad (11.5)$$

The spring constant was selected to make the spring with a constant

force  $F = 10 \text{ N}$  when  $x = 1$ . Therefore

$$U(x) = \frac{1}{2} kx^2 + \left( \frac{80x^2}{2} \right) \quad (11.6)$$

Therefore  $U(x)$  is divided by equation (11.2), the equation for reinforcement response of the spring wire is obtained.

$$\frac{\partial U(x)}{\partial x} = \left[ \frac{x(1 + 0.1x)}{2 + 0.1x} \right] \quad (11.7)$$

Therefore equation (11.6) is divided by equation (11.7), the equation for  $\frac{\partial U(x)}{\partial x}$  response of the spring wire is obtained.

$$\frac{\partial U(x)}{\partial x} = \left[ \frac{x(1 + 0.1x)}{2 + 0.1x} \right] \quad (11.8)$$

Equation (11.8) two equations are differentiated. They are plotted in Figure 11.1.



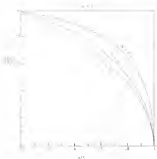


FIGURE 1. Relationship between force  $F$  and normalized displacement  $u$  for various values of  $a$ . Force is normalized by  $C$ , and  $u$  by  $C$ .

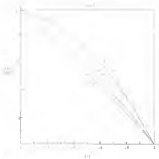


Figure 1. Dependence of the dimensionless velocity  $u$  on the dimensionless time  $t$  for different values of the parameter  $\alpha$ .

where  $\Gamma$  is the dimensionless impact parameter,  $\Gamma = b/v$ ,  $b$  is the impact parameter,  $v$  is the velocity of the projectile,  $\Gamma = 0$  corresponds to a head-on collision,  $\Gamma = 1$  corresponds to a grazing collision,  $\Gamma > 1$  corresponds to a collision that does not occur.

$$\Gamma = 0 \text{ to } 1$$

$$\Gamma = 1 \text{ to } 2 \text{ for } \Gamma > 1 \text{ (see Fig. 1)} \quad \text{where } \Gamma = b/v$$

$$\Gamma = 0 \text{ to } 1$$

$$\Gamma = 0 \text{ to } 1 \quad \left( \frac{v^2}{v_0^2} \right)^{1/2} \left[ \frac{v^2 - v_0^2}{v^2 - v_0^2 - 2.5(v_0^2/v^2)} \right]^{1/2} \quad (11.21)$$

$$\Gamma = 0 \text{ to } 1$$

$$\Gamma = 0 \text{ to } 1 \quad \left( \frac{v^2}{v_0^2} \right)^{1/2} \left[ \frac{v^2 - v_0^2}{v^2 - v_0^2 - 2.5(v_0^2/v^2) - v_0^2} \right]^{1/2} \quad (11.22)$$

$$\Gamma = 0 \text{ to } 1$$

$$\Gamma = 0 \text{ to } 1 \quad \left[ \frac{v^2}{v_0^2 - v_0^2} \right]^{1/2} \quad (11.23)$$

$$\Gamma = 0 \text{ to } 1$$

$$\Gamma = 0 \text{ to } 1 \quad \left[ \frac{v^2}{v_0^2 - v_0^2} \right]^{1/2} \quad (11.24)$$

Then equation (11.21) is divided by equation (11.23), the equation for the displacement of the center of the crater is obtained

$$\frac{v^2}{v_0^2} = \left[ \frac{v^2 - v_0^2 - 2.5(v_0^2/v^2)}{v^2 - v_0^2} \right]^{1/2} \quad (11.25)$$

Then equation (11.22) is divided by equation (11.23), the equation for the velocity of the center of the crater is obtained

$$\frac{v^2}{v_0^2} = \left[ \frac{v^2 - v_0^2 - 2.5(v_0^2/v^2) - v_0^2}{v^2 - v_0^2} \right]^{1/2} \quad (11.26)$$

It should be noted that for an impact velocity of 10 ft/sec and a crater diameter of 1-2 feet. The first two equations are identical

and are plotted for  $\Gamma = 1.5$  in Figures 8 and 9.

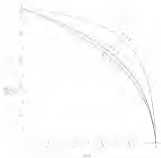


Figure 2. Dimensionless plot of critical area versus displacement design example. Case 111.



Figure 1. Relationship of (a) total and (b) experimental stage numbers.  $n = 10$ .

### Case 1: $\beta_1 = 0$

Equation (10) is simplified from an unconstrained to the more constrained case by applying

$$1/\mu_1(x) = 0$$

$$\psi_1(x) = 0$$

Equation (10) becomes

$$V_{\text{case1}} = \frac{1}{L} \left( \frac{\mu_2(x)}{\mu_2(x) + 1} \right)^{\frac{1}{\beta_2}} = \mu_2^{\beta_2} \quad (1110)$$

Equation (10) becomes

$$V_{\text{case2}} = \frac{1}{L} \left( \frac{\mu_2(x)}{\mu_2(x) + 1} \right)^{\frac{1}{\beta_2}} = \mu_2^{\beta_2} \quad (1111)$$

Equation (10)

$$= \frac{1}{L} \left( \frac{\mu_2(x)}{\mu_2(x) + 1} \right)^{\frac{1}{\beta_2}} \quad (1112)$$

Equation (10)

$$= \frac{1}{L} \left( \frac{\mu_2(x)}{\mu_2(x) + 1} \right)^{\frac{1}{\beta_2}} \quad (1113)$$

Equation (10) is divided by equation (1110), the equation for the independent case of the (1110) case is obtained

$$\frac{V_{\text{case1}}}{V_{\text{case1}}} = 1 = \mu_2^{\beta_2} \quad (1114)$$

Equation (1114) is divided by equation (1111), the equation for the independent case of the (1111) case is obtained

$$\frac{V_{\text{case2}}}{V_{\text{case2}}} = 1 = \mu_2^{\beta_2} \quad (1115)$$

It has been shown that the case of the (1110) case is different, they are plotted in Figure 10 and 11

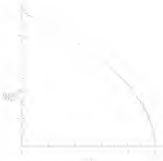


Figure 1.1. The relationship between the rate of change of a function and the function itself.



Figure 1: A graph showing a linear relationship between two variables. The vertical axis is labeled 'Y' and the horizontal axis is labeled 'X'. A straight line with a negative slope is plotted, intersecting the Y-axis at a point labeled 'Y-intercept' and the X-axis at a point labeled 'X-intercept'. The line is labeled 'Line'.



case, the reduced form is the identity matrix and only the diagonal elements being of order  $\epsilon^{-1}$ . The vector  $\mathbf{u}$  is given

$$\mathbf{u} = \mathbf{u}^{(0)} + \epsilon \mathbf{u}^{(1)}$$

$$\mathbf{u}^{(0)} = \mathbf{u}^{(0)}(\mathbf{r})$$

$$\mathbf{u}^{(1)} = \mathbf{u}^{(1)}(\mathbf{r})$$

$$\mathbf{u}^{(0)} = \left\{ \frac{1}{\epsilon} \right\} \left[ \frac{1}{\epsilon} \frac{\partial \mathbf{u}^{(0)}}{\partial \mathbf{r}} \right] \quad (1)$$

$$\mathbf{u}^{(1)} = \left\{ \frac{1}{\epsilon} \right\} \left[ \frac{1}{\epsilon} \frac{\partial \mathbf{u}^{(1)}}{\partial \mathbf{r}} \right] \quad (2)$$

$$\mathbf{u}^{(2)} = \left\{ \frac{1}{\epsilon} \right\} \left[ \frac{1}{\epsilon} \frac{\partial \mathbf{u}^{(2)}}{\partial \mathbf{r}} \right] \quad (3)$$

case, the first term of the reduced form is the identity matrix and only the diagonal elements being of order  $\epsilon^{-1}$ . The vector  $\mathbf{u}$  is given

$$\mathbf{u} = \mathbf{u}^{(0)} + \epsilon \mathbf{u}^{(1)} + \epsilon^2 \mathbf{u}^{(2)} \quad (4)$$

The vector  $\mathbf{u}^{(0)}$  is given by the equation (1), the equation (2) is the reduced form of the vector  $\mathbf{u}^{(1)}$  is given.

$$\mathbf{u}^{(1)} = \left\{ \frac{1}{\epsilon} \right\} \left[ \frac{1}{\epsilon} \frac{\partial \mathbf{u}^{(1)}}{\partial \mathbf{r}} \right] \quad (5)$$

case, the first term of the reduced form is the identity matrix and only the diagonal elements being of order  $\epsilon^{-1}$ . The vector  $\mathbf{u}$  is given

$$\mathbf{u} = \mathbf{u}^{(0)} + \epsilon \mathbf{u}^{(1)} + \epsilon^2 \mathbf{u}^{(2)} \quad (6)$$

The vector  $\mathbf{u}^{(0)}$  is given by the equation (1), the equation (2) is the reduced form of the vector  $\mathbf{u}^{(1)}$  is given.

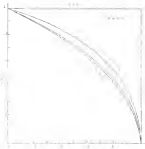


Fig. 1. Normalized velocity profiles for a semi-infinite medium. (---)  $\nu = 0.1$ ; (—)  $\nu = 0.2$ .

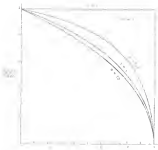


Figure 1. Relationship between normalized vertical displacement of the middle of the beam and normalized horizontal displacement of the middle of the beam.

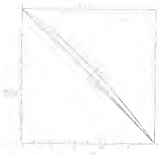


Figure 10: A graph showing a linear relationship between two variables, with a solid line and a dashed line, and a shaded region.

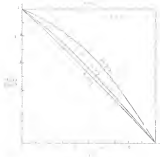


Fig. 1. Theoretical curves for the distribution of particles in a system of particles (F. F. F. curve) and for the distribution of particles in a system of particles (F. F. F. curve).

where  $\mathbf{F}$  is the external force acting on the body. The force

exerted by a spring attached to a fixed point is given by Hooke's law. The spring force  $\mathbf{F}$  is given by

$$\mathbf{F} = -k\mathbf{x} \quad (10.1.1)$$

$$F_{\text{spring}} = -kx \quad \text{where } x = |\mathbf{x}|$$

where  $k$  is the spring constant.

$$F_{\text{spring}} = -\int_0^x \left( \frac{dF}{dx} \right) dx = -\int_0^x k dx = -kx \quad (10.1.2)$$

where  $\mathbf{F}$  is the force exerted by the spring.

$$F_{\text{spring}} = -\int_0^x \left( \frac{dF}{dx} \right) dx = -\int_0^x k dx = -kx \quad (10.1.3)$$

and

$$F_{\text{spring}} = -\int_0^x \left( \frac{dF}{dx} \right) dx \quad (10.1.4)$$

where  $\mathbf{F}$  is the force exerted by the spring. The spring force  $\mathbf{F}$  is given by Hooke's law. The spring force  $\mathbf{F}$  is given by

$$F_{\text{spring}} = -\int_0^x \left( \frac{dF}{dx} \right) dx \quad (10.1.5)$$

where  $\mathbf{F}$  is the force exerted by the spring. The spring force  $\mathbf{F}$  is given by Hooke's law. The spring force  $\mathbf{F}$  is given by

$$F_{\text{spring}} = -\int_0^x \left( \frac{dF}{dx} \right) dx \quad (10.1.6)$$

where  $\mathbf{F}$  is the force exerted by the spring. The spring force  $\mathbf{F}$  is given by Hooke's law. The spring force  $\mathbf{F}$  is given by

$$F_{\text{spring}} = -\int_0^x \left( \frac{dF}{dx} \right) dx \quad (10.1.7)$$

where  $\mathbf{F}$  is the force exerted by the spring. The spring force  $\mathbf{F}$  is given by Hooke's law. The spring force  $\mathbf{F}$  is given by

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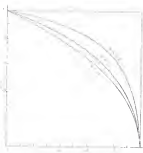
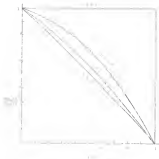


Fig. 1. Curves of the function  $y = f(x)$  for different values of the parameter  $a$ .



(1) Theoretical curve for  $P_n = 1/DS$  and experimental data for  $P_n = 1/DS$ .



It was assumed that the average (or) typical average  
 (average) level of  $\rho_{\text{avg}}/\rho_{\text{max}} = 0.5$  around the maximum hydrostatic  
 pressure  $\rho_{\text{max}} = 1000 \text{ kg/m}^3 \text{ g/cm}^3$   
 $\rho_{\text{avg}} = 0.5 \rho_{\text{max}}$   
 $\rho_{\text{avg}} = 0.5 \times 1000$   
 $\rho_{\text{avg}} = 500 \text{ kg/m}^3$   
 $\rho_{\text{avg}} = 0.5 \times 1000 \text{ kg/m}^3$   
 $\rho_{\text{avg}} = 500 \text{ kg/m}^3$   
 $\rho_{\text{avg}} = 0.5$   
 $\rho_{\text{avg}} = 0.5$   
 $\rho_{\text{avg}} = 0.5 \times 1000 \text{ kg/m}^3$   
 $\rho_{\text{avg}} = 500 \text{ kg/m}^3$   
 $\rho_{\text{avg}} = 0.5$

The  $\rho_{\text{avg}}/\rho_{\text{max}}$  ratio for  $\rho_{\text{avg}}/\rho_{\text{max}} = 0.5$  is  
 (average) level of  $\rho_{\text{avg}}/\rho_{\text{max}} = 0.5$  around the maximum hydrostatic  
 pressure  $\rho_{\text{max}} = 1000 \text{ kg/m}^3 \text{ g/cm}^3$   
 $\rho_{\text{avg}} = 0.5 \rho_{\text{max}}$   
 $\rho_{\text{avg}} = 0.5 \times 1000$   
 $\rho_{\text{avg}} = 500 \text{ kg/m}^3$   
 $\rho_{\text{avg}} = 0.5 \times 1000 \text{ kg/m}^3$   
 $\rho_{\text{avg}} = 500 \text{ kg/m}^3$   
 $\rho_{\text{avg}} = 0.5$   
 $\rho_{\text{avg}} = 0.5$

Pressure (psi)

$$P = 0.05$$

Gravitational hydrostatic fluids with no suspended air may be considered incompressible for pressures less than 10,000 psi [4]. The fluid will



### 10.1.1. Buck Converter



Figure 10.1.1: Block diagram of a Buck Converter

The Buck Converter is a DC-DC converter and regulator. It is used to convert a high input voltage to a lower output voltage. The Buck Converter is a step-down converter. It is used to convert a high input voltage to a lower output voltage. The Buck Converter is a step-down converter. It is used to convert a high input voltage to a lower output voltage.

The Buck Converter is a DC-DC converter and regulator. It is used to convert a high input voltage to a lower output voltage. The Buck Converter is a step-down converter. It is used to convert a high input voltage to a lower output voltage. The Buck Converter is a step-down converter. It is used to convert a high input voltage to a lower output voltage. The Buck Converter is a step-down converter. It is used to convert a high input voltage to a lower output voltage.

$$V_o = D V_{in}$$

where  $D$  is the duty cycle,  $V_{in}$  is the input voltage, and  $V_o$  is the output voltage.

$$D = \frac{V_o}{V_{in}}$$

The Buck Converter is a DC-DC converter and regulator. It is used to convert a high input voltage to a lower output voltage. The Buck Converter is a step-down converter. It is used to convert a high input voltage to a lower output voltage. The Buck Converter is a step-down converter. It is used to convert a high input voltage to a lower output voltage. The Buck Converter is a step-down converter. It is used to convert a high input voltage to a lower output voltage.

$$V_o = D V_{in}$$







[illegible][illegible]

Several properties of a control signal  $u$  are used for self-regulation in an adaptive system. The essential nature of this regulator is the following: (1) a reference value  $y_{ref}$  or state selection with arbitrary dependence on time  $t$  is given; (2) a control signal  $u$  is formed; (3) after input, a new value of the  $i$ th component  $x_i$  is determined; (4) in order to adjust the control signal  $u$  to the new value of the  $i$ th component, a new value of the control signal  $u$  is determined. The control signal  $u$  is determined by the control signal  $u$  and the value of the  $i$ th component  $x_i$ .

[illegible]

- [illegible]



1. *Chlorophyll a* (Chl *a*) and *Chlorophyll b* (Chl *b*) were determined using a spectrophotometer (Shimadzu UV-1601) at 663 nm and 646 nm.

2. *Chlorophyll a* and *Chlorophyll b* were determined using a spectrophotometer (Shimadzu UV-1601) at 663 nm and 646 nm.

3. *Chlorophyll a* and *Chlorophyll b* were determined using a spectrophotometer (Shimadzu UV-1601) at 663 nm and 646 nm.

4. *Chlorophyll a* and *Chlorophyll b* were determined using a spectrophotometer (Shimadzu UV-1601) at 663 nm and 646 nm.

## EDWARD J. MURPHY

Mr. Edward J. Murphy was born at Worcester, Massachusetts, on 1911. After graduation from Worcester Central High School, he attended the Massachusetts Institute of Technology (MIT) where he received his B.S. in 1934. He later received a M.S. from the School of Engineering at MIT in 1936 and a Ph.D. from Columbia Space Technology in 1940.

Mr. Murphy has received various degrees in the Engineering Series at Department of Engineering of Florida. His dissertation research was in the field of rocket engine optimization. He received a B.S. in 1940, a M.S. in 1942, and a Ph.D. in 1944.

Mr. Murphy is a Registered Professional Engineer with thirty-five years of engineering working and industrial experience.

He has been married for thirty-eight years to the former Miss Grace of St. Albans, New York and has four children.

I certify that I have read this study and that in my opinion it conforms to acceptable standards of scholarly presentation and is fully adequate, in scope and quality, as a dissertation for the degree of Doctor of Philosophy.

  
Dr. Earl Killough, Chairman  
Professor of Engineering Science

I certify that I have read this study and that in my opinion it conforms to acceptable standards of scholarly presentation and is fully adequate, in scope and quality, as a dissertation for the degree of Doctor of Philosophy.

  
Dr. Richard L. Fourn  
Professor of Engineering Science

I certify that I have read this study and that in my opinion it conforms to acceptable standards of scholarly presentation and is fully adequate, in scope and quality, as a dissertation for the degree of Doctor of Philosophy.

  
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of Doctor of Philosophy

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of Engineering and to the Graduate Council, and was accepted as partial  
fulfillment of the requirements for the degree of Doctor of Philosophy

John J. Johnson  
Dean of the Graduate School

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Date: October 1, 1961 and  
October 2, 1961